## **GENERAL NOTES REVISIONS** (UNLESS OTHER WISE SPECIFIED) ZONE REV DESCRIPTION DATE **APPROVED** Fabricate per AC 43.13. Assembly shop practice per AC 43.13. 2011-0X Initial Release J.BROWN Flange corner radii are to be R .25. Alodine all aluminum parts per Mil-C-5541. Epoxy prime per Mil-P-23377. 6. Remove all burrs and sharp edges. Torque per AC 43.13-1A or MIL-N-25027. Surface finish of parts to be 125 microinches Install safety wire per AC 43.13-1A or MS33540. All dimensions are prior to finish applications. Equally spaced attachment tolerance to be +/- 0.100". 12. Minimum attachment spacing to be 4D+.060" center to center. Attachments located from part edge 2D+.060" for details, 2D minimum on assembly, 2D from joggles and adjacks structure. Identify per MIL-STD-130 using rubber stamp or and written with indelible ink. Small assemblies may be tagged. Symbols + between end fastener symbols indicate identical fasteners equally spaced. Install solid rivets per AC 43.13-1A. Install all other fasteners per manufacturer's instructions. 17. Faying surface seal all external skin mating joints and install fasteners wet using PRX431G or P/S 800C seafant. 18. Fillet or brush coat edges of mating parts using PR1422-B2, CS3204 class B, or P/S\890 class B sealant. 19. Mix all sealants and adhesives per the Mfg's instructions. 20. Antenna must be bonded to airframe for P-static and lightning protection. Reference AC 43.13-1B Chapter 12 Section 2 Paragraph 12-9 (b) for inspection antenna guideline & Chapter 11 Section 15 Paragraph 11-193 for lightning protection. Also reference a/c manufacturer's maintenance manual for recommended antenna bonding practices. **INSTALLATION NOTES** Install the A-33 Antenna at FS xxx BL 0.0 per the manufacturer's instructions. Doubler may be larger, and fastener spacing greater than shown. Flanges may be added. Nuts and washers may be used in place of nutplates. Install item 4 in item 2 only, NOT in skin. This drawing has been prepared for a specific alteration to a specific aircraft or component based on information supplied to Aviation Engineering Consultants of USA, LLC (AECUSA-LLC) by the person responsible for arranging the alteration or his agent. The use of the data on any other aircraft or component may lead to an unsafe alteration and therefore the data is not to be used for any other purpose than for which it was originally supplied without the written approval of AECUSA-LLC. Screw 33 560-949-01 Antenna <u>6</u> 5 NAS1097AD4 AR Rive MS21062-3 8 Rivet & MŚ20059− <u>@</u> Nutplate 0.040 2024 - T3. AMS-QQ - A - 250/5 $0.040 \times 5 \times 4$ Doubler nstallation -01 NOTE PART NO. NOMENCLATURE OR DESCRIPTION MATERIAL/SPECIFICATION STOCK SIZE PARTS LIST NOTICE — This drawing is the property of Aviation Engineering Consultants of USA, LLC, and any disclosure of its contents to others is given in confidence. It must not be reproduced DIMENSIONS ARE IN INCHES AVIATION ENGINEERING TOLERANCES: TITLE: A \$33 GPS ANTENNA INST CONSULTANTS of USA, LLC .XX= ±.03 .XXX= ±.015 or copied in whole or in part or used in any way except with specific approval and permission of the General Managi Cape Coral, FL Aviation Engineering Consultants of USA, LLC APPRVD DWG. NO. 00-1500-03-04-xxx ANGLES ±0°-30° 2011-0X 7/25/2011 1:05:25 PM



